

## FIFTH SEMESTER B.TECH (INSTRUMENTATION AND CONTROL ENGINEERING)

END SEMESTER EXAMINATIONS, NOV/DEC 2015

SUBJECT: SYSTEM MODELING AND SIMULATION [ICE 319]

Time: 3 Hours

MAX. MARKS: 50

### Instructions to Candidates:

- ❖ Answer **ANY FIVE FULL** questions.
- ❖ Missing data may be suitably assumed.

- 1A. Define transfer function? State whether transfer function modeling technique is applicable to non-linear system and whether the transfer function is independent of the input of a system. List the drawbacks in the conventional transfer function model. 4
- 1B. A common example of a two-input control system is a home shower with separate valves for hot and cold water. The objective is to obtain (1) a desired temperature of the shower water and (2) a desired flow of water. Sketch a block diagram of the closed-loop control system. 3
- 1C. Obtain the transfer function  $X_2(s)/U(s)$  for the mechanical system shown in Fig Q(1C). 3
- 2A. Write the differential equation governing the mechanical system shown in Fig Q(2A). Draw the force-current analogous circuit and verify by writing node equation. 5
- 2B. Derive transfer function and state space model of an armature controlled DC motor. 5
- 3A. In the liquid-level system of Fig. Q(3A) assume that the outflow rate  $Q$  m<sup>3</sup>/sec through the outflow valve is related to the head  $H$  m by  $Q = 0.01\sqrt{H}$ . Also assume that when the inflow rate  $Q_i$  is 0.015 m<sup>3</sup>/sec the head stays constant. For  $t < 0$  the system is at steady state at ( $Q_i=0.015$  m<sup>3</sup>/sec). At  $t=0$  the inflow valve is closed and so there is no inflow for  $t \geq 0$ . Find the time necessary to empty the tank to half the original head. The capacitance  $C$  of the tank is 2 m<sup>2</sup>. 5
- 3B. For the hydraulic linear actuator shown in Fig Q (3B), obtain the transfer function  $Y(s)/X(s)$ . List the advantages and disadvantages of hydraulic system. 5
- 4A. Define transmissibility for force excitation and motion excitation. Explain the variation of transmissibility with  $\beta(=\frac{\omega}{\omega_n})$  and damping factor. 5
- 4B. For the mechanical system shown in Fig Q (4B), when  $P = 1$  N and  $\omega = 2$  rad/s, the steady state amplitude of  $x(t)$  is found to be 0.05 m. If the forcing frequency is changed to  $\omega = 10$  rad/s, the steady-state amplitude of  $x(t)$  is found to be 0.02 m. Determine the values of  $b$  and  $k$ . 5
- 5A. With a schematic diagram, briefly explain about longitudinal and lateral – directional motion of an aircraft. Derive moment equations of an aircraft. 7
- 5B. Obtain the state space model for the single axis space craft as shown in Fig Q(5B), flying near-earth or an interplanetary trajectory. Moments acting on the spacecraft are control moment  $M_C(t)$  and disturbance moment  $M_D(t)$ . 3

- 6A. Consider the inverted – pendulum system shown in Fig Q(6A). Assume that the mass of the inverted pendulum is  $m$ . The center of gravity of the pendulum is located at the center of the rod. Assuming that  $\theta$  is small, derive the differential equation describing the whole system. 5
- 6B. Write about the physical setup of a ball and beam system. Considering both rotational and translational motion of the ball, derive the differential equation governing the system. 5

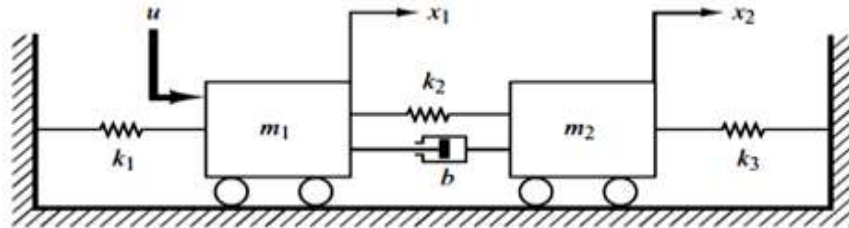


Fig. Q (1C)

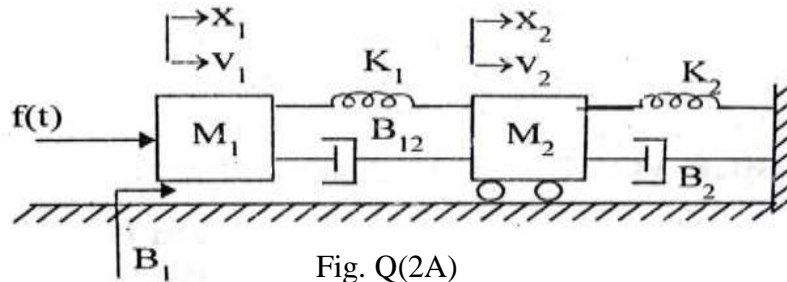


Fig. Q(2A)

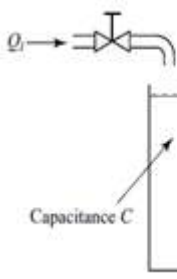


Fig Q(3A)

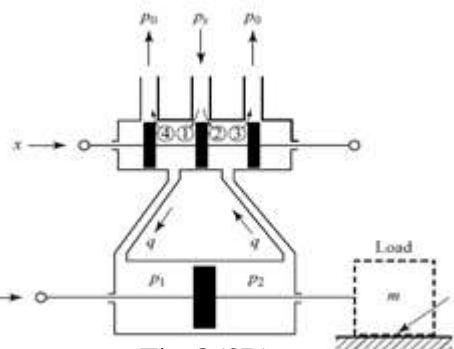


Fig Q(3B)

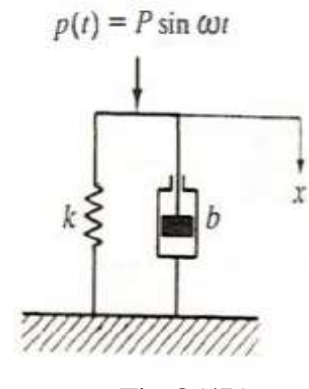


Fig Q(4B)

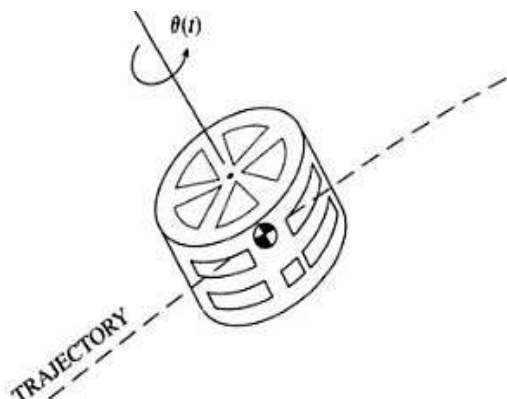


Fig Q(5B)

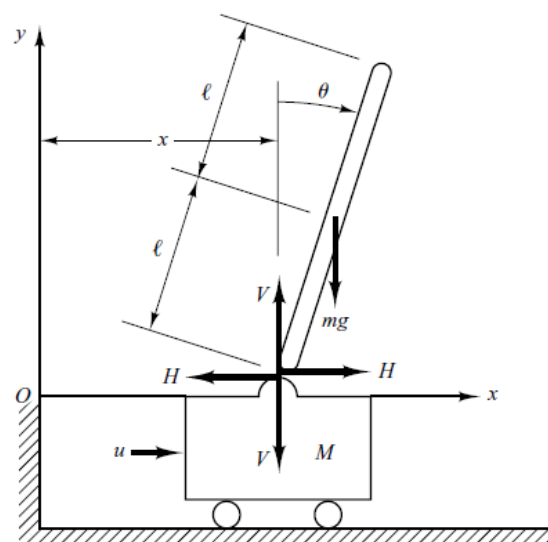


Fig Q(6A)