

FIRST SEMESTER M.TECH. (AEROSPACE ENGINEERING) **END SEMESTER EXAMINATIONS, NOV/DEC 2016**

SUBJECT: NAVIGATION AND GUIDANCE OF AEROSPACE VEHICLES [ICE 5104]

Time: 3 Hours

MAX. MARKS: 50

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Instructions to Candidates:

- ✤ Answer ALL the questions.
- ✤ Missing data may be suitably assumed.
- 1A. An aircraft is flying on a straight-line leg of 73 Nautical miles between two points with a 5 course of 215 degree as measured using a map. Local magnetic deviation is (-11^{0}) . Aircraft will fly at 12000 feet with a true air speed of 165 knots. Aircraft is flying in air that has wind from 350° at 25 knots. Calculate the time of flying.
- 1B. With diagram, explain NGC loop. List the applications of navigation, guidance and 3 control systems
- **1C.** Write about following navigation methods: (i) Dead Reckoning (ii) Celestial Navigation
- 2A. Explain inertial navigation system. What is stable platform INU and strapdown INU. List 5 the advantages of strapdown system.
- **2B.** Position vector P_A (1, 2, 3) of an aircraft, represented in ECI frame (frame A) need to be 5 transformed to body frame (frame B). The orientation of A and B co-ordinate frames are as shown in figure (2B). Perform the transformation using following techniques:
 - Euler angle transformation (i)
 - (ii) **Ouaternions**
- **3A.** Write about integrated navigation system. What are the various types of navigational aids 5 available for integrated navigation system?
- Briefly explain the working of radio based navigation aid VOR/DME. 3 **3B**.
- **3C.** Obtain the scale factor for a SDFG rate gyro.

- **4A.** Write about following guidance systems:
 - (i) Inertial guidance
 - (ii) Homing guidance
 - (iii) Commanded guidance
- 4B. With diagram, explain the working of a Quartz-Flex accelerometer
 5A. Explain proportional navigation guidance. With intercept geometry, derive the generalized
 5 equation governing PN guidance.
- **5B.** With block diagram, explain decoupled and integrated GNC system architecture. **3**
- **5C.** Briefly explain, roll stabilization system for a missile



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