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MANIPAL INSTITUTE OF TECHNOLOGY
MANIPAL
A Constituent Institution of Manipal University

V SEMESTER B.TECH. (AERONAUTICAL ENGINEERING)

END SEMESTER EXAMINATIONS, NOV/DEC 2016

SUBJECT: ORBITAL MECHANICS [AAE 4012]

**REVISED CREDIT SYSTEM
(05/01/2017)**

Time: 3 Hours

MAX. MARKS: 50

Instructions to Candidates:

- ❖ Answer **ALL** the questions.
- ❖ Missing data may be suitable assumed.

- 1A.** What are the contributions of Galileo for the development of Celestial Mechanics? **(2)**
- 1B.** State and prove Kepler's third law **(3)**
- 1C.** Calculate the altitude and speed of a geostationary earth satellite. **(5)**
- 2A.** How a satellite can be shifted from an initial orbit to another orbit of different eccentricity and inclination? **(2)**
- 2B.** Explain Hohmann transfer with necessary diagrams. **(3)**
- 2C.** Find the total delta-v requirement for a bi-elliptic transfer from a geocentric circular orbit of 7000 km radius to one of 105,000 km radius. Let the apogee of the first ellipse be 210,000 km. **(5)**
- 3A.** Explain synodic time. **(2)**
- 3B.** What do you mean by Sphere Of Influence (SOI) of a planet? **(3)**
- 3C.** Calculate the minimum wait time for initiating a return trip from Mars to earth. Radius of earth is 149.6×10^6 Km. Radius of Mars is 227.9×10^6 Km. μ of sun is 132.7×10^9 units **(5)**
- 4A.** What is orbit determination process? Explain various methods involved. **(2)**

- 4B.** Explain orbital elements with proper diagram. (3)
- 4C.** How position and velocity vectors of a satellite is determined using Lagrange co-efficients. (5)
- 5A.** What are the consequences of orbital perturbations? (2)
- 5B.** Explain earth centered inertial coordinate system with proper diagram (3)
- 5C.** Explain various methods employed for space debris removal. (5)