



## VII SEMESTER B.TECH. END SEM (Additional OPE) EXAMINATIONS (MAR 2021)

### SUBJECT: DESIGN AND DRAWING OF CHEMICAL EQUIPMENT [CHE 4102]

#### REVISED CREDIT SYSTEM

Date of Exam: 19/03/2021

Time of Exam: 2 – 5 pm

Time: 3 Hours

Max. Marks: 50

#### Instructions to Candidates:

- ❖ Answer ALL the questions & missing data may be suitably assumed
- ❖ Use of code book/data book/formula sheet are permitted (ATTACHED)

|   |   |         |
|---|---|---------|
| 1 | <p>Determine the thickness of the reaction vessel and the jacket from the following data:<br/> Vessel shell internal diameter: 2130 mm; Jacket length: 2500 mm; Width of channel jacket: 100 mm; Internal pressure (Shell): <math>0.35 \text{ N/mm}^2</math>; Internal pressure (Jacket): <math>0.55 \text{ N/mm}^2</math>; Design temperature: <math>150^\circ\text{C}</math>; Allowable stress value at design temperature: <math>96 \text{ N/mm}^2</math>; Spot radiography given.<br/> Dished head and bottom – Internal diameter: 2130 mm; Crown radius : 2130 mm; Knuckle radius: 128 mm</p>  | 5 + 5   |
| 2 | <ul style="list-style-type: none"> <li>Determine the maximum operating pressure for the vessel shown based on the shell dimensions and material.</li> <li>For this pressure select a suitable gasket material and determine the optimum number of bolts (20 mm), minimum gasket width and flange thickness. The service temperature is <math>400^\circ\text{F}</math>.</li> </ul> <div data-bbox="534 1086 1316 1904"> </div> <p><b>Hint:</b> For pressures above 20 bar, ring-joint gasket (<math>m=5.5</math>; <math>y=124 \text{ N/mm}^2</math>) can be chosen. For SA-212, Grade B : <math>f = 1.19 \times 10^8 \text{ N/m}^2</math> ; For SA-193, Grade B : <math>f = 1.36 \times 10^9 \text{ N/m}^2</math>; For SA-105, Grade B : <math>f = 1.19 \times 10^9 \text{ N/m}^2</math></p> | 2.5 x 4 |

|                 |   |                     |
|-----------------|---|---------------------|
| <p><b>3</b></p> | <p>80,000 kg/h distilled water (shell) has to be cooled from 34°C to 29.5°C in a 1 – 2 STHE. Raw water (tube) enters at 24°C and the exit temperature should not exceed 30°C. Use Chrome steel (1% Cr), 16 BWG, ¾” OD tube (OD =1.905cm; ID=1.575cm), 1” triangular pitch, number of tubes = 302, shell ID = 21.25”, and baffle spacing = 40% of shell ID. The fouling resistances (m<sup>2</sup> K/W) are 8.81x10<sup>-5</sup> and 1.76 x 10<sup>-4</sup> for distilled and raw water respectively. Assume a correction factor (F<sub>T</sub>) of 0.85 for multi-pass.</p> <ol style="list-style-type: none"> <li>the tube side heat transfer coefficient</li> <li>the shell side heat transfer coefficient</li> <li>the length of the tubes required.</li> </ol> <p><b>Useful Formula:</b><br/> <b>Tube Side:</b> <math>a_t = (\pi/4 d_i^2) N/n</math>; <math>Nu = 0.027 Re^{0.8} Pr^{0.33}</math><br/> <b>Shell side:</b> <math>a_s = D_s C B / P_T</math>; <math>Nu = 0.36 Re^{0.55} Pr^{0.33}</math><br/> <math>D_{eq} = 4\{[0.44 P_T^2 - (\pi d_o^2/8)] / [\pi d_o/2]\}</math></p>  | <p><b>4+4+2</b></p> |
| <p><b>4</b></p> | <p>A water-cooled, 1–1 shell-and-tube freon condenser with in-tube condensation of R-22 @37°C (<math>c_{pL} = 1.305</math> kJ/kg K; <math>v_L = 8.3734 \times 10^{-4}</math> m<sup>3</sup>/kg; <math>v_g = 0.01643</math> m<sup>3</sup>/kg; <math>\mu_L = 1.86 \times 10^{-4}</math> Pa.s; <math>\mu_g = 1.39 \times 10^{-5}</math> Pa.s; <math>k_L = 0.082</math> W/m K; <math>\lambda = 169</math> kJ/kg; <math>Pr = 2.96</math>) has to be designed. City water (Inlet and outlet temperatures are 18°C &amp; 26°C respectively) is used as solvent. The physical properties at the average temperature of the coolant are: <math>c_{pL} = 4.181</math> kJ/kg K; <math>\mu_L = 959 \times 10^{-6}</math> Pa.s; <math>k_L = 0.606</math> W/m K; <math>Pr = 6.61</math>. Fouling resistance: 1.76x10<sup>-4</sup> m<sup>2</sup> K/W for both inside and outside.</p> <p><b>Design parameters:</b><br/> Design cooling load: 100 kW; One tube pass; Pitch: 1” Square; Shell dia: 15.25”; Baffle Spacing: 35 cm; Number of Tubes: 137; Size of tubes: 0.75” OD &amp; 0.68” ID; Vapor quality = 50%.</p> <ol style="list-style-type: none"> <li>Determine the shell side heat transfer coefficient.</li> <li>By using <i>Cavallini and Zecchin</i> theory, calculate the tube side heat transfer coefficient. Take vapor quality = 50%</li> <li>Calculate the length of the condenser.</li> </ol> | <p><b>4+4+2</b></p> |
| <p><b>5</b></p> | <p>A forward-feed evaporator (triple effect) is being used to evaporate a solution containing 5wt% solids to a concentrated solution of 50wt%. Saturated steam at 8.5 atm is being used. The feed rate is 2x10<sup>5</sup> kg/hr at 16°C. Take <math>U_1 = 4500</math>, <math>U_2 = 2800</math> and <math>U_3 = 1700</math> W/m<sup>2</sup>K; <math>\Delta T_1 = 28</math> K, <math>\Delta T_2 = 23</math> K and <math>\Delta T_3 = 42</math> K. Calculate the surface area of the evaporator and the steam economy.</p> <p>Determine,</p> <ol style="list-style-type: none"> <li>the amount of concentrated liquor leaving each evaporator</li> <li>the surface area of the evaporator and</li> <li>the steam economy</li> </ol>   | <p><b>5+3+2</b></p> |

## FORMULA SHEET / DATA SHEET

### CHE 4102 Design and Drawing of Chemical Process Equipment

#### 1. Internal Pressure Vessels – Minimum thickness for various shapes

|   |   |
|---|---|
| Cylinder: $t = \frac{PD_i}{2fJ-P} = \frac{PD_o}{2fJ+P}$   | $t$ : min thickness of the shell plates exclusive of corrosion allowance<br>$P$ : design pressure<br>$D_i$ : inside diameter of the shell<br>$D_o$ : outside diameter of the shell<br>$f$ : allowable stress value<br>$J$ : joint efficiency factor |
| Sphere: $t = \frac{PD_i}{4fJ-P} = \frac{PD_o}{4fJ+P}$   |   |
| Hemi Sphere: $t = \frac{PD_i}{4fJ}$   |   |
| Flat plate: $t = C D_e \sqrt{\frac{P}{f}}$  | $C$ : a design constant, dependent on the edge constraint<br>$D_e$ : nominal plate diameter   |
| Ellipsoid: $t = \frac{PD_i C}{2fJ}$ & $C = \frac{1}{4} [2 + K^2]$                                   | $K$ : ratio of major to minor axis  |
| Tori-sphere: $t = \frac{PD_i C}{2fJ}$ & $C = \frac{1}{4} \left[ 3 + \sqrt{\frac{R_c}{R_k}} \right]$ | $R_c$ : crown radius<br>$R_k$ : knuckle radius  |
| Conical: $t = \frac{PD_i}{2fJ-P} \left( \frac{1}{\cos \alpha} \right)$                              | $\alpha$ : Half-cone angle  |

#### 2. Volume of various shapes

|   |                                      |
|---|--------------------------------------|
| Tori-sphere: $V = 0.0809 D_i^3$                       | $D_i$ : inside diameter of the shell |
| Ellipsoid: $V = \frac{\pi}{24} D_i^3$                 |                                      |
| Conical: $V = \frac{\pi h}{12} [D_i^2 + D_i d + d^2]$ | $d$ : dia of the small end           |

#### 3. External pressure vessels

|   |   |
|---|---|
| Allowable working external pressure:<br>$P_a = \frac{B}{14.22(D_o/t)} \text{ kgf/cm}^2$ | $t$ : min thickness of the shell plates exclusive of corrosion allowance<br>$D_o$ : outside diameter of the shell<br>$B$ : Factor B from Chart (Fig F. 2 – Indian Standard Code for unfired pressure vessel, BIS 2825-1969) |
|---|---|

#### 4. Flange and Gasket design

|  |   |
|--|---|
| Actual width of gasket, $N = (G_o - G_i)/2$  | $G_o$ : outside diameter of the gasket<br>$G_i$ : inside diameter of the gasket |
| Ratio of gasket diameters, $\left( \frac{G_o}{G_i} \right) = \left( \frac{y - mP}{y - P(m+1)} \right)^{0.5}$ | $P$ : design pressure<br>$y$ : gasket seating stress<br>$m$ : gasket factor     |

|   |   |
|---|---|
|   |   |
| Bolt load due to initial gasket load reaction,<br>$W_{m1} = \pi b G y$  | $b_0$ : Basic gasket seating width (before applying load) = $N/2$<br>$b$ : Effective gasket seating width (after applying load)<br><ul style="list-style-type: none"> <li>– <math>b = b_0</math>, when <math>b_0 \leq 6.3\text{mm}</math></li> <li>– <math>b = \frac{1}{2} \sqrt{b_0}</math>, when <math>b_0 &gt; 6.3\text{mm}</math></li> </ul> $G$ : diameter at location of gasket load reaction<br>= mean diameter of gasket contact face if $b_0 \leq 6.3\text{mm}$<br>= (Inside diameter of gasket) + $2N - 2b$ if $b_0 > 6.3\text{mm}$ |
| Bolt load at operating conditions, $W_{m2} = H + H_p$<br>$H$ = Hydrostatic end force = $(\pi/4) G^2 P$<br>$H_p$ = total joint contact surface compression = $\pi(2b) G m P$ |   |
| Minimum bolt area, $A_{m1} = (W_{m1} / f_a)$ and $A_{m2} = (W_{m2} / f_b)$  | $f_a$ & $f_b$ : allowable stress values at atmospheric and operating conditions respectively<br>Bolt Circle Diameter, $B$ = OD of gasket + 2(bolt diameter) + 12mm  |
| Minimum gasket width, $N_{\min} = (A_b f_a) / 2\pi y G$   |   |
| $h_G$ = radial distance from gasket load reaction to the bolt circle = $(B - G)/2$  |   |
| Thickness of the flange, $t_f = G \left[ \frac{P}{K_f} \right]^{0.5}$ &<br>$K = \frac{1}{0.3 + \frac{1.5(W_m)(h_G)}{(H)(G)}}$   |   |

## 5. Process Design of Shell and Tube Heat Exchanger

|  |   |
|--|---|
| Heat Duty, $Q = (\dot{m} C_p \Delta T)_{\text{hot}} = (\dot{m} C_p \Delta T)_{\text{cold}}$  | $\dot{m}$ : mass flowrate<br>$C_p$ : specific heat capacity<br>$\Delta T$ : Temperature difference<br>Hot : hot stream<br>Cold: cold stream |
| Total area required, $A_o = (Q / U_{OD} \Delta T_{LMTD})$  | $U_{OD}$ : Overall heat transfer coefficient including dirt factor<br>$\Delta T_{LMTD}$ : Log mean temperature difference                   |
| Number of tubes, $N = A_o / \pi d_o L$   | $L$ = length of the tubes<br>$d_o$ = outside diameter of the tubes  |
| Equivalent diameter, $D_{eq}$<br>Triangular Pitch: $D_{eq} = 4 \{ [0.44 P_T^2 - (\pi d_o^2/8)] / [\pi d_o/2] \}$<br>Square Pitch: $D_{eq} = 4 \{ [P_T^2 - (\pi d_o^2/4)] / [\pi d_o] \}$ | $P_T$ = pitch<br>$d_o$ = outside diameter of the tubes  |
| Tube side cross sectional area, $a_t = (\pi/4 d_i^2) N/n$  | $N$ : number of tubes<br>$n$ : number of tube passes  |
| Shell area available for flow, $a_s = D_s C B / P_T$   | $D_s$ : shell diameter<br>$B$ : Baffle spacing  |

|  |   |
|--|---|
|  | <p>C : clearance, = <math>P_T - d_o</math><br/> <math>P_T</math> = pitch<br/> <math>d_o</math> = outside diameter of the tubes</p>  |
| <p>Tube Side: <math>Nu = 0.027 Re^{0.8} Pr^{0.33}</math><br/> Shell side: <math>Nu = 0.36 Re^{0.55} Pr^{0.33}</math></p>   | <p>Nu = Nusselt number<br/> Re = Reynolds number<br/> Pr = Prandl number</p>  |
| <p>Clean overall heat transfer Coefficient, <math>U_{OC}</math><br/> <math>[1/U_{OC}] = [1/h_o] + [d_o/d_i][1/h_i]</math></p>  | <p><math>h_o</math> = individual heat transfer coefficient (outside)<br/> <math>h_i</math> = individual heat transfer coefficient (inside)<br/> <math>d_i</math> = inside diameter of the tubes<br/> <math>d_o</math> = outside diameter of the tubes</p>   |
| <p>Overall heat transfer coefficient including dirt factor<br/> <math>[1/U_{OD}] = [1/U_{OC}] + [1/h_{di}] + [1/h_{do}]</math></p>   | <p><math>h_{di}</math> : heat transfer coefficient for deposit (tubes)<br/> <math>h_{do}</math> : heat transfer coefficient for deposit (shell)</p>   |
| <p>Shell side pressure drop, <math>\Delta P_s</math><br/> <math display="block">\Delta P_s = \frac{f G_s^2 D_s L}{2 \times 10^6 D_e s B} \quad (kN/m^2)</math></p>           | <p><math>f</math> = friction factor = <math>1.87 (Re_s)^{-0.2}</math><br/> <math>Re_s</math> = Shell side Reynolds number<br/> <math>G_s</math> = mass velocity<br/> <math>D_s</math> = shell diameter<br/> <math>D_e</math> = equivalent diameter of shell in m<br/> <math>L</math> = length of the tube<br/> <math>B</math> = baffle spacing<br/> <math>s</math> = specific gravity</p> |
| <p>Tube side pressure drop, <math>\Delta P_t</math><br/> <math display="block">\Delta P_t = \frac{f G_t^2 L n}{2 \times 10^6 d_i s} + 2.26 n v^2 s \quad (kN/m^2)</math></p> | <p><math>f</math> = friction factor = <math>0.72 (Re_t)^{-0.33}</math><br/> <math>Re_t</math> = Tube side Reynolds number<br/> <math>n</math> = number of passes<br/> <math>L</math> = length of the tube<br/> <math>G_t</math> = mass velocity<br/> <math>d_i</math> = tube diameter<br/> <math>s</math> = specific gravity</p>  |

## 6. Process Design of Condenser

|   |   |
|---|---|
| <p>Nusselt's Theory (Laminar Flow):<br/> <math display="block">Nu = \frac{h_m d}{k_l} = 0.728 \left[ \frac{\rho_l (\rho_l - \rho_g) g \lambda d^3}{\mu_l (T_{sat} - T_w) k_l} \right]^{1/4}</math></p>        | <p><math>h_m</math> : mean condensation film coefficient<br/> <math>d</math> : diameter of tube<br/> <math>\rho_l</math>: condensate density<br/> <math>k_l</math>: thermal conductivity<br/> <math>\mu_l</math>: condensate viscosity<br/> <math>g</math> : gravitational acceleration<br/> <math>\lambda</math> : latent heat of condensation<br/> <math>x</math>: is the vapor quality, the mass fraction of vapor<br/> <math>G</math> : Mass velocity</p> |
| <p>Travis theory (Turbulent):<br/> <math display="block">Re_l = \frac{G(1-x)d}{\mu_l}</math> <math display="block">Nu = \frac{h_{TP} d}{k_l} = Pr_l Re_l^{0.9} \frac{F_l(X_{tt})}{F_2(Re_l, Pr_l)}</math></p> |   |

Non-dimensional parameter,  $F_1$ , is

$$F_1 = 0.15 \left[ \frac{1}{X_{tt}} + \frac{2.85}{X_{tt}^{0.476}} \right]$$

$X_{tt}$  is the Lockhart – Martinelli parameter

$$X_{tt} = \left( \frac{1-x}{x} \right)^{0.9} \left( \frac{\rho_g}{\rho_l} \right)^{0.5} \left( \frac{\mu_l}{\mu_g} \right)^{0.1}$$

$$F_2 = 0.7 Pr_l Re_l^{0.5} \text{ for } Re_l \leq 50$$

$$F_2 = 5 Pr_l + 5 \ln[1 + Pr_l(0.09636 Re_l^{0.585} - 1)] \text{ for } 50 < Re_l \leq 1125$$

$$F_2 = 5 Pr_l + 5 \ln(1 + 5 Pr_l) + 2.5 \ln(0.00313 Re_l^{0.812}) \text{ for } Re_l > 1125$$

Cavallini and Zecchin theory (Turbulent):

$$h_{TP} = 0.05 Re_{eq}^{0.8} Pr_l^{0.33} \frac{k_l}{d}$$

$$Re_{eq} = Re_v \left[ \frac{\mu_v}{\mu_l} \right] \left[ \frac{\rho_l}{\rho_v} \right]^{0.5} + Re_l$$

$$Re_l = \frac{G(1-x)d}{\mu_l}$$

$$Re_v = \frac{Gxd}{\mu_v}$$

Shaw theory (Turbulent):

$$h_{TP} = h_l \left[ 1 + \frac{3.8}{Z^{0.95}} \right]$$

$$Z = \left[ \frac{1-x}{x} \right]^{0.8} Pr^{0.4}$$

$$h_l = 0.023 \left[ \frac{G(1-x)d}{\mu_l} \right]^{0.8} \frac{Pr^{0.4} k}{d}$$



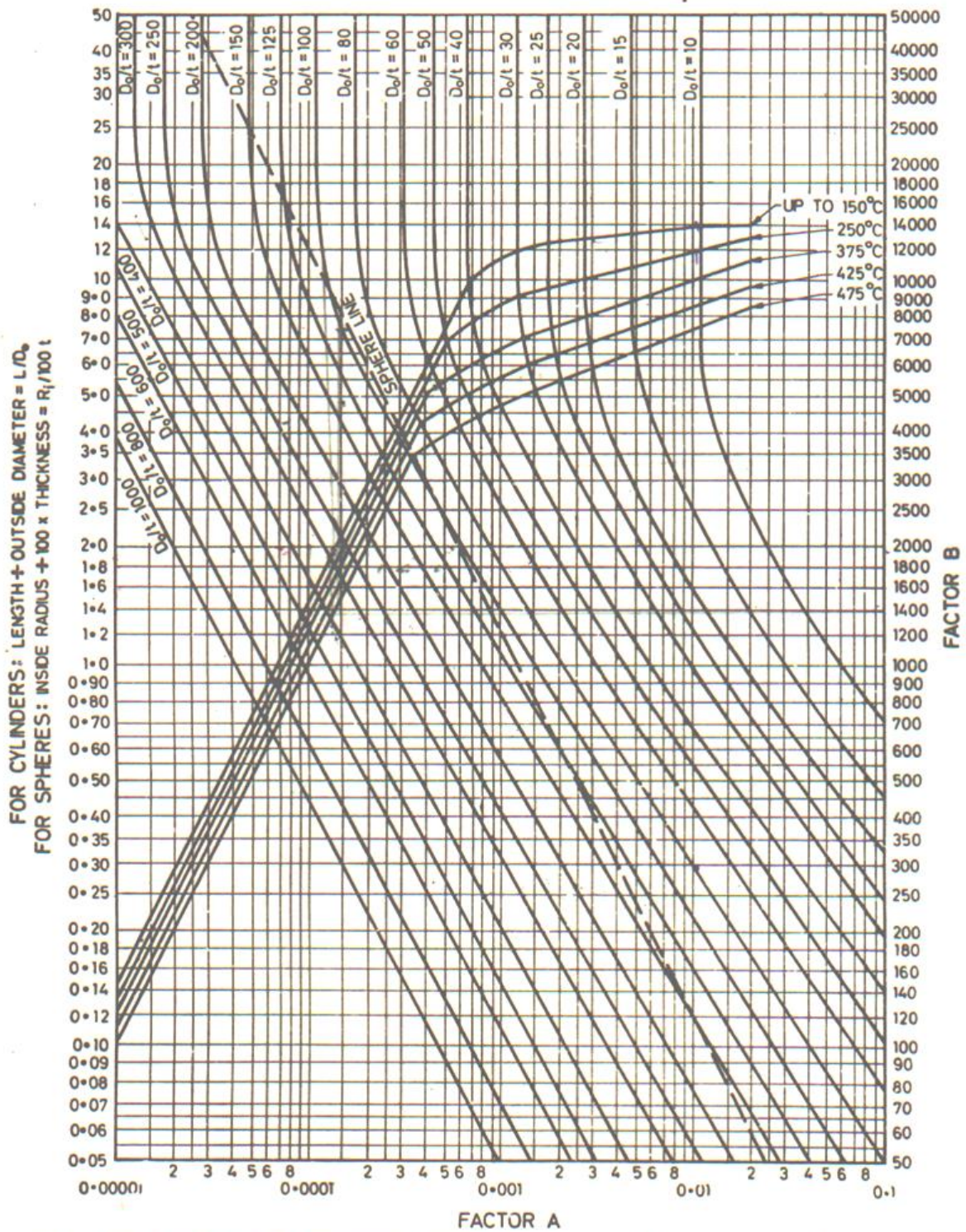
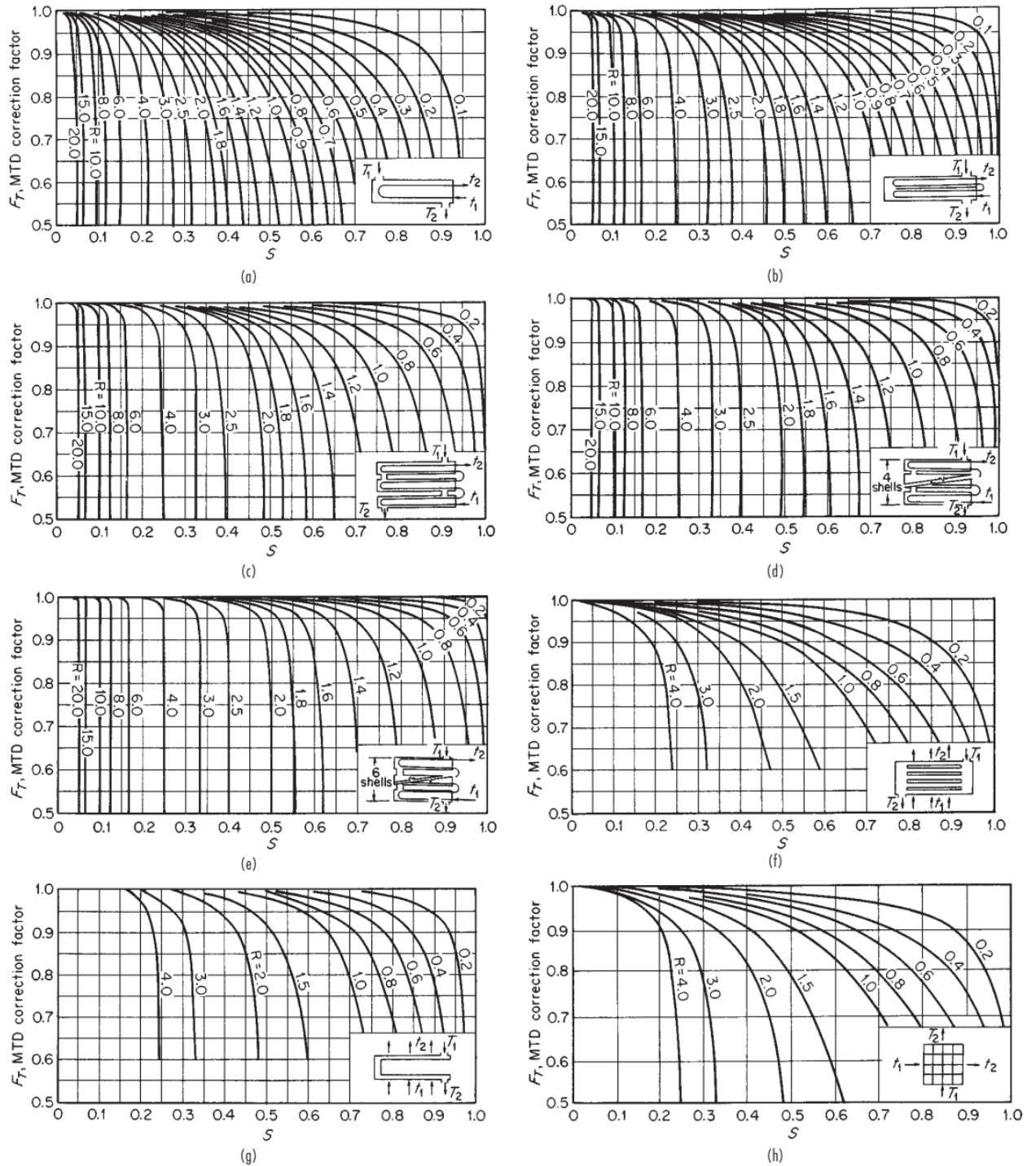


FIG. F.2 CHARTS FOR DETERMINING SHELL THICKNESSES OF CYLINDRICAL AND SPHERICAL VESSELS UNDER EXTERNAL PRESSURE, CARBON AND LOW ALLOY STEEL

## 11-6 HEAT-TRANSFER EQUIPMENT



**FIG. 11-4** LMTD correction factors for heat exchangers. In all charts,  $R = (T_1 - T_2)/(t_2 - t_1)$  and  $S = (t_2 - t_1)/(T_1 - t_1)$ . (a) One shell pass, two or more tube passes. (b) Two shell passes, four or more tube passes. (c) Three shell passes, six or more tube passes. (d) Four shell passes, eight or more tube passes. (e) Six shell passes, twelve or more tube passes. (f) Cross-flow, one shell pass, one or more parallel rows of tubes. (g) Cross-flow, two passes, two rows of tubes; for more than two passes, use  $F_T = 1.0$ . (h) Cross-flow, one shell pass, one tube pass, both fluids unmixed



**A.2-9 Properties of Saturated Steam and Water (Steam Table),  
SI Units**

| Temperature<br>(°C) | Vapor<br>Pressure<br>(kPa) | Specific Volume<br>(m <sup>3</sup> /kg) |             | Enthalpy<br>(kJ/kg) |             | Entropy<br>(kJ/kg · K) |             |
|---------------------|----------------------------|---|-------------|---------------------|-------------|------------------------|-------------|
|                     |                            | Liquid                                  | Sat'd Vapor | Liquid              | Sat'd Vapor | Liquid                 | Sat'd Vapor |
| 0.01                | 0.6113                     | 0.0010002                               | 206.136     | 0.00                | 2501.4      | 0.0000                 | 9.1562      |
| 3                   | 0.7577                     | 0.0010001                               | 168.132     | 12.57               | 2506.9      | 0.0457                 | 9.0773      |
| 6                   | 0.9349                     | 0.0010001                               | 137.734     | 25.20               | 2512.4      | 0.0912                 | 9.0003      |
| 9                   | 1.1477                     | 0.0010003                               | 113.386     | 37.80               | 2517.9      | 0.1362                 | 8.9253      |
| 12                  | 1.4022                     | 0.0010005                               | 93.784      | 50.41               | 2523.4      | 0.1806                 | 8.8524      |
| 15                  | 1.7051                     | 0.0010009                               | 77.926      | 62.99               | 2528.9      | 0.2245                 | 8.7814      |
| 18                  | 2.0640                     | 0.0010014                               | 65.038      | 75.58               | 2534.4      | 0.2679                 | 8.7123      |
| 21                  | 2.487                      | 0.0010020                               | 54.514      | 88.14               | 2539.9      | 0.3109                 | 8.6450      |
| 24                  | 2.985                      | 0.0010027                               | 45.883      | 100.70              | 2545.4      | 0.3534                 | 8.5794      |
| 25                  | 3.169                      | 0.0010029                               | 43.360      | 104.89              | 2547.2      | 0.3674                 | 8.5580      |
| 27                  | 3.567                      | 0.0010035                               | 38.774      | 113.25              | 2550.8      | 0.3954                 | 8.5156      |
| 30                  | 4.246                      | 0.0010043                               | 32.894      | 125.79              | 2556.3      | 0.4369                 | 8.4533      |
| 33                  | 5.034                      | 0.0010053                               | 28.011      | 138.33              | 2561.7      | 0.4781                 | 8.3927      |
| 36                  | 5.947                      | 0.0010063                               | 23.940      | 150.86              | 2567.1      | 0.5188                 | 8.3336      |
| 40                  | 7.384                      | 0.0010078                               | 19.523      | 167.57              | 2574.3      | 0.5725                 | 8.2570      |
| 45                  | 9.593                      | 0.0010099                               | 15.258      | 188.45              | 2583.2      | 0.6387                 | 8.1648      |
| 50                  | 12.349                     | 0.0010121                               | 12.032      | 209.33              | 2592.1      | 0.7038                 | 8.0763      |
| 55                  | 15.758                     | 0.0010146                               | 9.568       | 230.23              | 2600.9      | 0.7679                 | 7.9913      |
| 60                  | 19.940                     | 0.0010172                               | 7.671       | 251.13              | 2609.6      | 0.8312                 | 7.9096      |
| 65                  | 25.03                      | 0.0010199                               | 6.197       | 272.06              | 2618.3      | 0.8935                 | 7.8310      |
| 70                  | 31.19                      | 0.0010228                               | 5.042       | 292.98              | 2626.8      | 0.9549                 | 7.7553      |
| 75                  | 38.58                      | 0.0010259                               | 4.131       | 313.93              | 2635.3      | 1.0155                 | 7.6824      |
| 80                  | 47.39                      | 0.0010291                               | 3.407       | 334.91              | 2643.7      | 1.0753                 | 7.6122      |
| 85                  | 57.83                      | 0.0010325                               | 2.828       | 355.90              | 2651.9      | 1.1343                 | 7.5445      |
| 90                  | 70.14                      | 0.0010360                               | 2.361       | 376.92              | 2660.1      | 1.1925                 | 7.4791      |
| 95                  | 84.55                      | 0.0010397                               | 1.9819      | 397.96              | 2668.1      | 1.2500                 | 7.4159      |
| 100                 | 101.35                     | 0.0010435                               | 1.6729      | 419.04              | 2676.1      | 1.3069                 | 7.3549      |

## A.2-9 SI Units, Continued

| Temperature<br>(°C) | Vapor<br>Pressure<br>(kPa) | Specific Volume<br>(m <sup>3</sup> /kg) |             | Enthalpy<br>(kJ/kg) |             | Entropy<br>(kJ/kg · K) |             |
|---------------------|----------------------------|---|-------------|---------------------|-------------|------------------------|-------------|
|                     |                            | Liquid                                  | Sat'd Vapor | Liquid              | Sat'd Vapor | Liquid                 | Sat'd Vapor |
| 105                 | 120.82                     | 0.0010475                               | 1.4194      | 440.15              | 2683.8      | 1.3630                 | 7.2958      |
| 110                 | 143.27                     | 0.0010516                               | 1.2102      | 461.30              | 2691.5      | 1.4185                 | 7.2387      |
| 115                 | 169.06                     | 0.0010559                               | 1.0366      | 482.48              | 2699.0      | 1.4734                 | 7.1833      |
| 120                 | 198.53                     | 0.0010603                               | 0.8919      | 503.71              | 2706.3      | 1.5276                 | 7.1296      |
| 125                 | 232.1                      | 0.0010649                               | 0.7706      | 524.99              | 2713.5      | 1.5813                 | 7.0775      |
| 130                 | 270.1                      | 0.0010697                               | 0.6685      | 546.31              | 2720.5      | 1.6344                 | 7.0269      |
| 135                 | 313.0                      | 0.0010746                               | 0.5822      | 567.69              | 2727.3      | 1.6870                 | 6.9777      |
| 140                 | 316.3                      | 0.0010797                               | 0.5089      | 589.13              | 2733.9      | 1.7391                 | 6.9299      |
| 145                 | 415.4                      | 0.0010850                               | 0.4463      | 610.63              | 2740.3      | 1.7907                 | 6.8833      |
| 150                 | 475.8                      | 0.0010905                               | 0.3928      | 632.20              | 2746.5      | 1.8418                 | 6.8379      |
| 155                 | 543.1                      | 0.0010961                               | 0.3468      | 653.84              | 2752.4      | 1.8925                 | 6.7935      |
| 160                 | 617.8                      | 0.0011020                               | 0.3071      | 675.55              | 2758.1      | 1.9427                 | 6.7502      |
| 165                 | 700.5                      | 0.0011080                               | 0.2727      | 697.34              | 2763.5      | 1.9925                 | 6.7078      |
| 170                 | 791.7                      | 0.0011143                               | 0.2428      | 719.21              | 2768.7      | 2.0419                 | 6.6663      |
| 175                 | 892.0                      | 0.0011207                               | 0.2168      | 741.17              | 2773.6      | 2.0909                 | 6.6256      |
| 180                 | 1002.1                     | 0.0011274                               | 0.19405     | 763.22              | 2778.2      | 2.1396                 | 6.5857      |
| 190                 | 1254.4                     | 0.0011414                               | 0.15654     | 807.62              | 2786.4      | 2.2359                 | 6.5079      |
| 200                 | 1553.8                     | 0.0011565                               | 0.12736     | 852.45              | 2793.2      | 2.3309                 | 6.4323      |
| 225                 | 2548                       | 0.0011992                               | 0.07849     | 966.78              | 2803.3      | 2.5639                 | 6.2503      |
| 250                 | 3973                       | 0.0012512                               | 0.05013     | 1085.36             | 2801.5      | 2.7927                 | 6.0730      |
| 275                 | 5942                       | 0.0013168                               | 0.03279     | 1210.07             | 2785.0      | 3.0208                 | 5.8938      |
| 300                 | 8581                       | 0.0010436                               | 0.02167     | 1344.0              | 2749.0      | 3.2534                 | 5.7045      |

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